



MODULE SPECIFICATION

Part 1: Information			
Module Title	Aero-Elasticity		
Module Code	UFMEWC-15-M	Level	Level 7
For implementation from	2018-19		
UWE Credit Rating	15	ECTS Credit Rating	7.5
Faculty	Faculty of Environment & Technology	Field	Engineering, Design and Mathematics
Department	FET Dept of Engin Design & Mathematics		
Contributes towards			
Module type:	Standard		
Pre-requisites	Aero Structures 2018-19, Aerodynamics and Flight 2018-19, Pilot Studies and Aerodynamics 2018-19		
Excluded Combinations	None		
Co- requisites	None		
Module Entry requirements	None		

Part 2: Description
<p>Overview: Module Entry requirements: The module is intended for science and engineering graduates, or equivalent, with strong mathematical skills.</p> <p>Pre-requisites: students must take UFMFX6-15-2 Aerostructures and either: UFMFY6-30-2 Aerodynamics and Flight or UFMF9C-30-2 Pilot studies & Aerodynamics</p> <p>Educational Aims: See learning outcomes.</p> <p>Outline Syllabus: Introduction: Flutter, Aero-elasticity, Modes, Properties of stiffness matrices, freefree, applying constraints, Fixed root modes, Eigen solution Orthogonality of normal modes and transformation of multi-degree of freedom systems into modal equations, Free-free modes</p> <p>Basic considerations: Wing inertial and flexural axes, Control surfaces, Static divergence, Control reversal, Influence on design, Strip theory – single element, Unsteady aerodynamics, Theodorsen, Minhinnick, frequency parameters, Aerodynamic stiffness and damping, Structural</p>

STUDENT AND ACADEMIC SERVICES

damping, 2 D.O.F. flutter equation, Classical equations to predict flutter speeds, Addition of control surfaces

Multi-strip fixed root wing bending torsion flutter: Stiffness properties, Calculation of wing modes and inertias, Orthogonal transform of mass and stiffness matrices to obtain modal set, 5 strip wing aerodynamics, Interpolate mode shapes onto strips, Assemble aerodynamic stiffness and damping, Flutter solution – needs flutter solution algorithm available, Effects of mass, flexural axis, frequency parameter, density, Types of flutter solution, matched and unmatched frequency parameter

Control surface flutter: How controls work, Attachment and control stiffness, Add freedom to wing model, Control surface flutter derivatives

Laboratory demonstration of free-free modes: Simple free-free beam, Shake test, Symmetric/asymmetric modes, Mode measurement and plotting

Free-free modes flutter: Symmetric/asymmetric, Symmetric flutter – pitch, vertical freedom and bending torsion, Comparison of mode shapes/inertias with fixed wing calculations earlier, Flutter analysis

Airframe modal characteristics: Wings, Tails, Engines, Weapons/stores, Complete airframe, Measuring modes, Representing free-free modes, Shake tests methods, Back to free-free model – prediction of forced response to shaker input – student exercise

Flutter testing: Purpose, Design requirements, Single, multiple failures, Store combinations, Excitation methods, Analysis methods, Telemetry, Flight envelope, Achieving speeds, Critical parts of envelope, Control system failure cases, Safety.

Teaching and Learning Methods: Students will learn through a combination of formal lectures and tutorials sessions.

Part 3: Assessment

The module is examined via an exam of 3 hours which will cover the taught issues.

First Sit Components	Final Assessment	Element weighting	Description
Examination - Component A	✓	100 %	Exam (180 minutes)
Resit Components	Final Assessment	Element weighting	Description
Examination - Component A	✓	100 %	Exam (180 minutes)

STUDENT AND ACADEMIC SERVICES

Part 4: Teaching and Learning Methods		
Learning Outcomes	On successful completion of this module students will be able to:	
	Module Learning Outcomes	
	MO1	Key principles of aero-elasticity, theoretical considerations coupled with experiments
	MO2	The aspects of flutter
	MO3	The classical equations to predict flutter
	MO4	The experimental methods and analysis of flutter
	MO5	The physics of aero-elasticity for an aircraft and its components
	MO6	The effects of mass, flexural axis, frequency parameter, types of flutter solutions
	MO7	The numerical/experimental data from a control surface
	MO8	Calculations of the free-free mode flutter
	MO9	Design requirements including flutter
	MO10	Modelling of a control surface flutter
	MO11	The practical issues of dynamic measurements and analysis and testing
	MO12	Awareness of professional literature
	MO13	Problem formulation and decision making
MO14	Self-management skills	
Contact Hours	Contact Hours	
	Independent Study Hours:	
	Independent study/self-guided study	114
	Total Independent Study Hours:	114
	Scheduled Learning and Teaching Hours:	
	Face-to-face learning	36
	Total Scheduled Learning and Teaching Hours:	36
	Hours to be allocated	150
	Allocated Hours	150
	Reading List	<p>The reading list for this module can be accessed via the following link:</p> <p>https://uwe.rl.talis.com/modules/ufmewc-15-m.html</p>